International

P.O. BOX 3090 ALBANY, GEORGIA 31706-3090 PHONE 229/883-1440 FAX 229/439-9790

Custom Kit CK-AG-1 February 10, 1975

GOODYEAR DISC BRAKE INSTALLATION

EFFECTIVITY: S-2R, S/N 1416R thru 1938R.

APPROVAL: FAA DER Approved

ELECTRICAL LOAD: Not Applicable.

WEIGHT AND BALANCE: The weight and balance change resulting from installation of this

Custom Kit is as follow:

Weight (LBS) H-ARM (Inches) H-Moment (In-Lbs)

-1.4 3.1 -4.0

SUPPLY DATA: Parts required to comply with this publication are furnished in Kit

form and consist of the following:

QTY	PART NO.	DESCRIPTION
2 ea	50151-1	Brake Assy.
2 ea	50365-13	Axle
2 ea	50365-9	Weld Assy
2 ea	AN815-4	Union
2 ea	S-0309-012	O-Ring
2 ea	8.50 – 10	Inner Tube (Type III, Style "S")
2 ea	8.50 - 10	Tire (10 Ply rated, Type III LP)
16 ea	AN5C11A	Bolt
2 ea	AN7502-16	Nut
16 ea	AN960C516	Washer
2 ea	MS21258-C16	Washer
16 ea	MS21043-5	Nut
2 ea	MS24665-306	Cotter Pin
2 ea	9531395	Wheel Assy
	CONSISTING OF:	
	1 9511135	Dust Cap
	1 95-2882	Retainer Ring
	1 95-3035	Retainer Ring
	2 95-3083	Seal Retainer
	1 95-3087	Felt Ring
	1 07100	Bearing
	1 19150	Bearing
1 ea	No. CK-AG-01	Custom Kit

PAGE 1 of 4 CK-AG-1

NOTE

If any problems are encountered while installing this custom kit, contact General Aviation Division, Rockwell International, Customer Service Department, Bethany, Oklahoma 73008.

INSTRUCTIONS:

- Release parking brakes.
- B. Jack aircraft as outlined in Section III of Thrush Commander Owner's Manual.
- C. Remove and discard existing left and right wheel assemblies.
- D. Disconnect and cap brake hydraulic line at brake housing.
- E. Remove and discard existing left and right brake assemblies
- F. Remove and discard existing left and right axles.
- G. Using 50365-9 weld assembly as template, insert –9 weld assembly in existing landing gear strut assembly and scribe diameter of –9 weld assembly on existing brake attach plate (see Figure 1.).
- H. Remove 50365-9 weld assembly from strut assembly and cut existing brake attach plate to 4.70 inch diameter (see Figure 1.).
- I. Reinsert 50365-9 weld assembly in landing gear strut assembly and drill eight (8) 0.3150 (±0.035) inch diameter holes in existing brake attach plate (see Figure 1.).
- J. Temporarily bolt 50365-9 weld assembly to brake attach plate, insert 53065-13 axle into –9 weld assembly and drill a 0.313 (±0.001) inch diameter hole in weld assembly and axle through existing hole in strut assembly (see Figure 1.).
- K. Remove axle and weld assembly from strut assembly and deburr all holes and rough edges.
- L. Install 50365-9 weld assembly and 50365-13 axle on landing gear strut assembly using existing hardware (see Figure 1.).
- M. Install 50151-1 brake assembly on brake attach plate and reconnect existing brake line to brake assembly (see Figure 1.).

NOTE

Hand form existing brake line as necessary for proper fit.

N. Install tube and tire on 9531395 wheel assembly. Torque wheel half bolts to 18 foot-pounds (see Figure 1.).

NOTE

Tire to be inflated to 38 psi.

O. Install 19150 bearing, 95-3083 seal retainer (2 places), 95-3087 ring felt in inner wheel half with 95-3035 retainer ring (see Figure 1.).

PAGE 2 of 4 CK-AG-1

- P. Install 9531395 wheel assembly and 07100 bearing on main landing gear axle with AN7502-16 nut, MS21258-C16 washer and MS24665-306 cotter pin as follows:
 - 1. While rotating wheel, torque nut to 80 inch pounds
 - 2. While rotating wheel, back nut off to zero (0) inch pounds.
 - 3. While rotating wheel, finger tighten nut not to exceed 40 inch pounds.
 - 4. If nut is not locking position, advance not to next position (not to exceed 30° and/or 40 inchpounds) while wheel is rotating.
 - 5. Install MS24665-306 cotter pin.
- Q. Drain master brake cylinders and fill with NAPA Super Heavy 9-2 hydraulic brake fluid.
- R. Remove any air in brake line by successive pumping of brake pedal.
- S. Remove aircraft from jacks.

AIRCRAFT RECORDS: Make appropriate entry in airplane permanent maintenance records as follows:

Custom Kit N0. CK-AG-1 dated February 10, 1975, entitled "Goodyear Disc Brake Installation", accomplished (date) .

PAGE 3 of 4 CK-AG-1

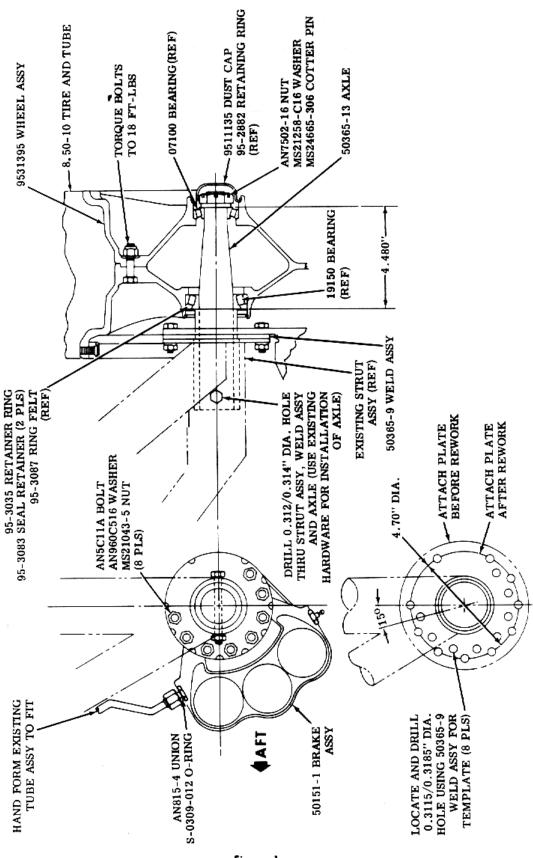


Figure 1.

PAGE 4 of 4 CK-AG-1